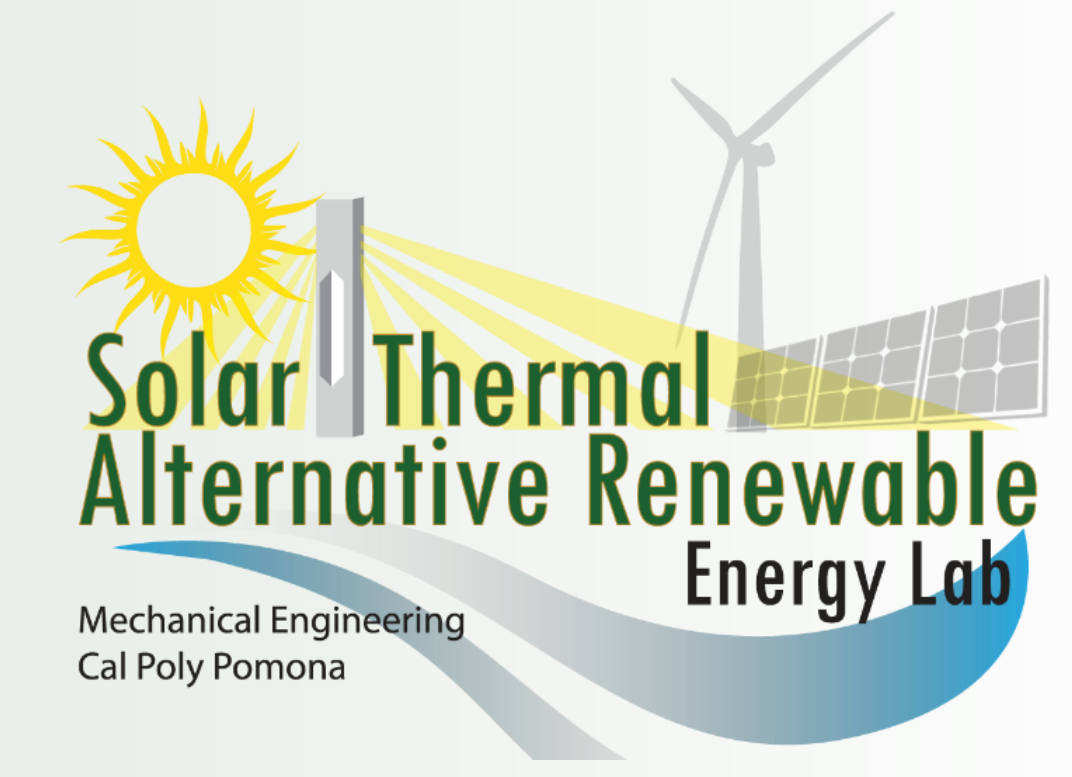


Actively Cooled Venus Lander Instrument Payload Using a Multi-Cascade Refrigeration Cycle

by

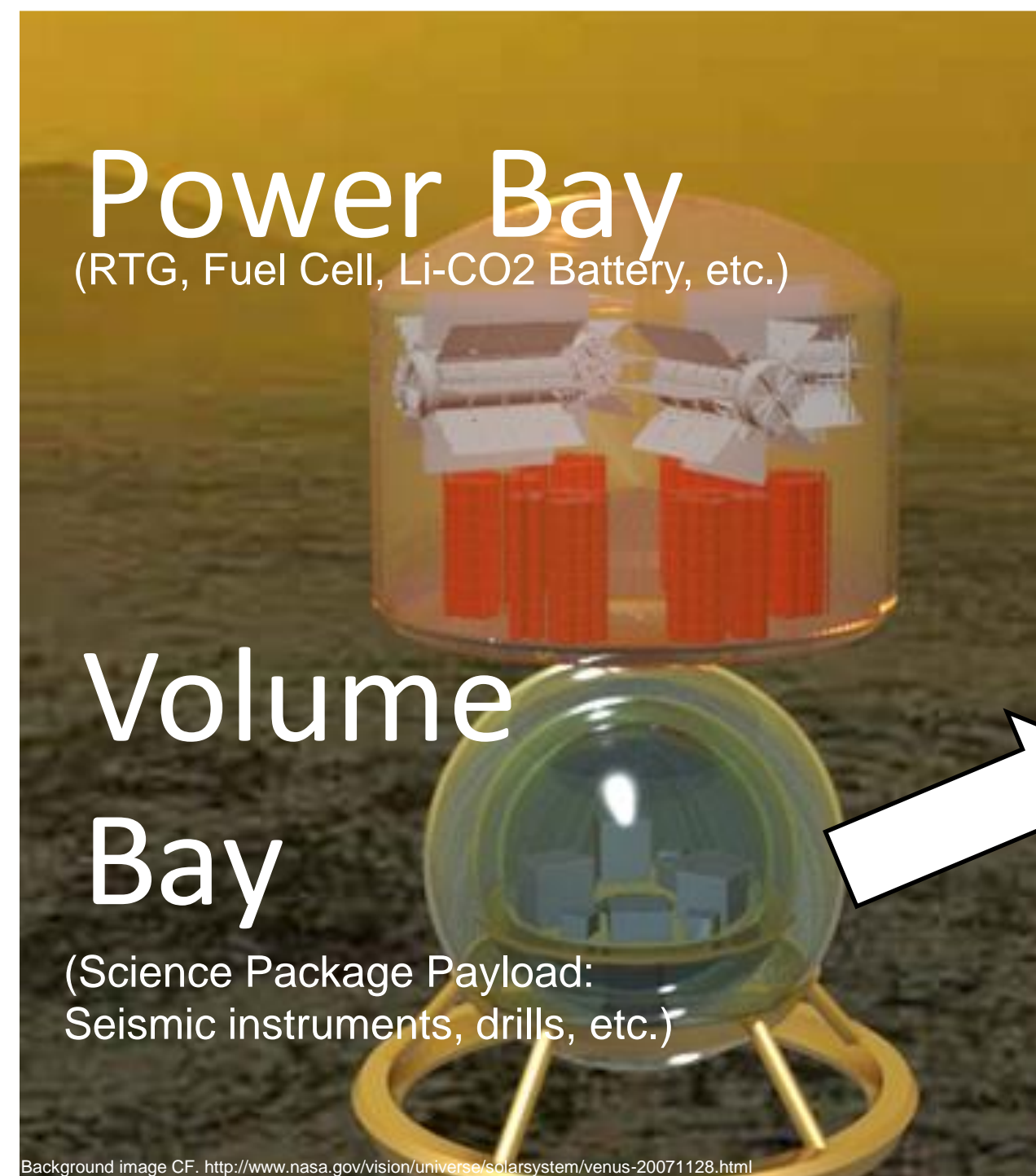
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²Ingenium Technical Services, Inc., ³Orbital/ATK



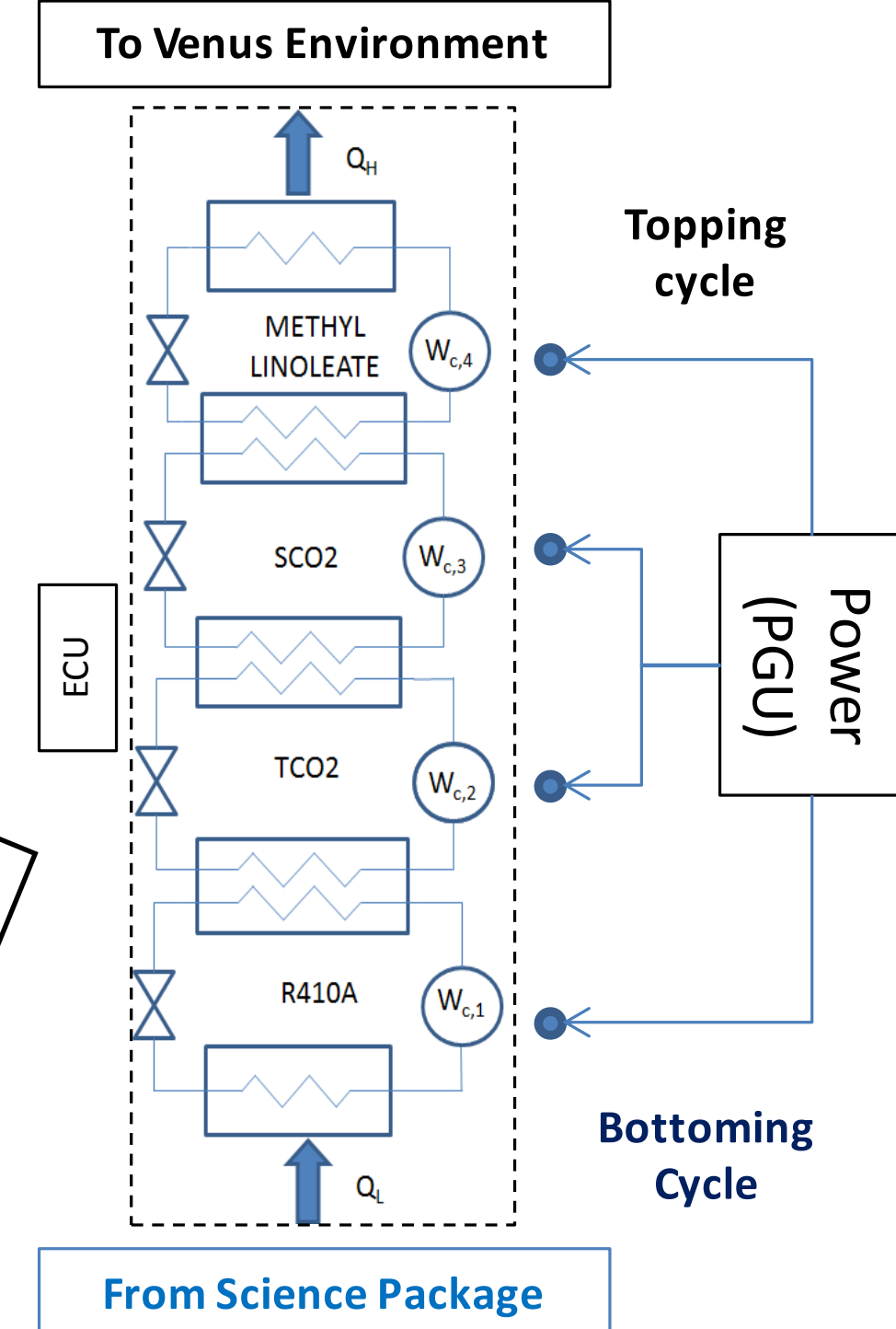
presented at
 The 48th Lunar & Planetary Science Conference (LPSC), March 20-14, 2017, The Woodlands, TX
 Research project funding to Calif. State Polytechnic Univ. at Pomona provided by Ingenium Technical Services, Inc.
 Project research in collaboration with Orbital / ATK

Background

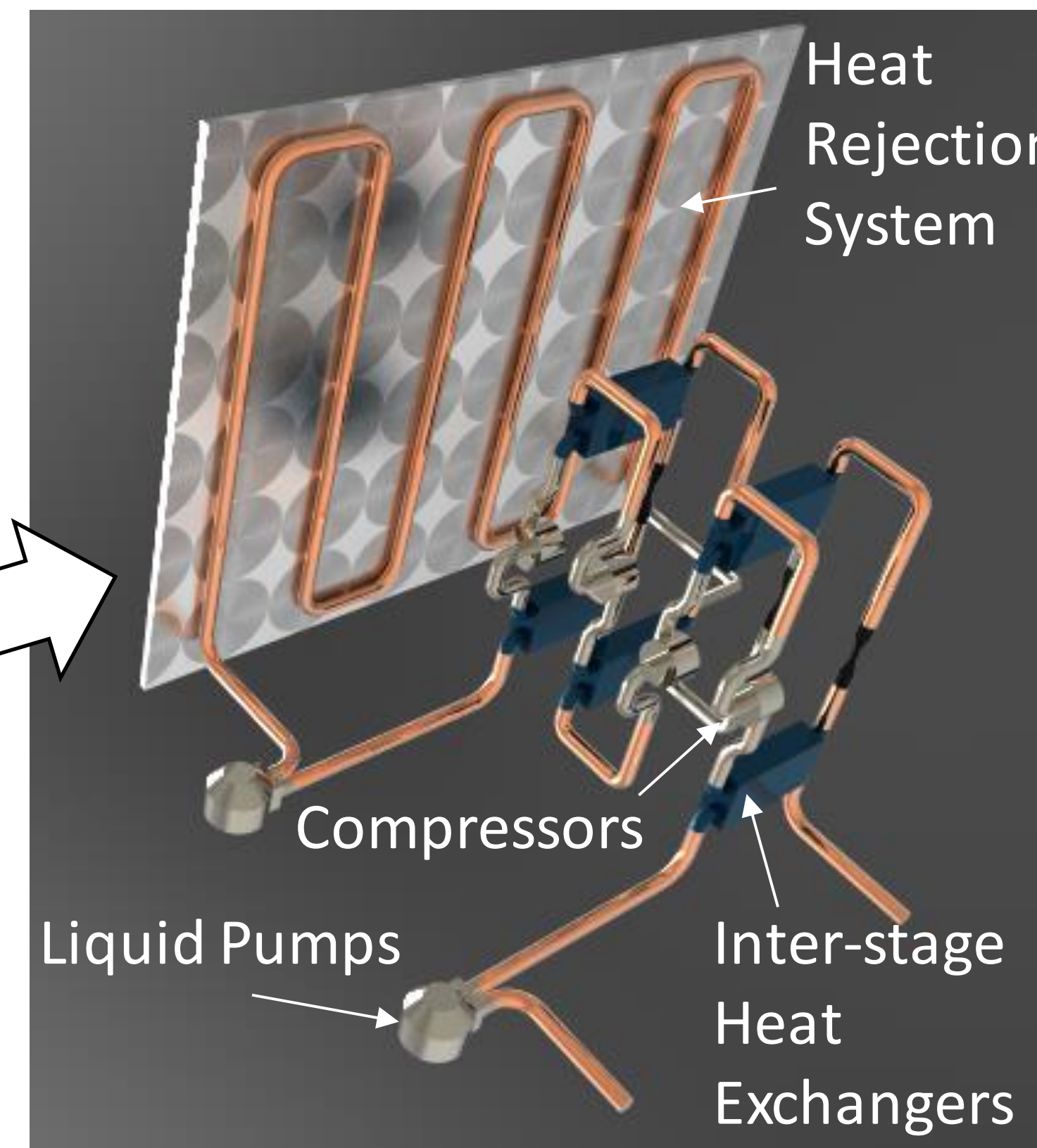
- Use of a cascade refrigeration cycle to actively cool Venus Lander payload electronics to extend mission life on the harsh Venus surface



Cascade cycle schematic

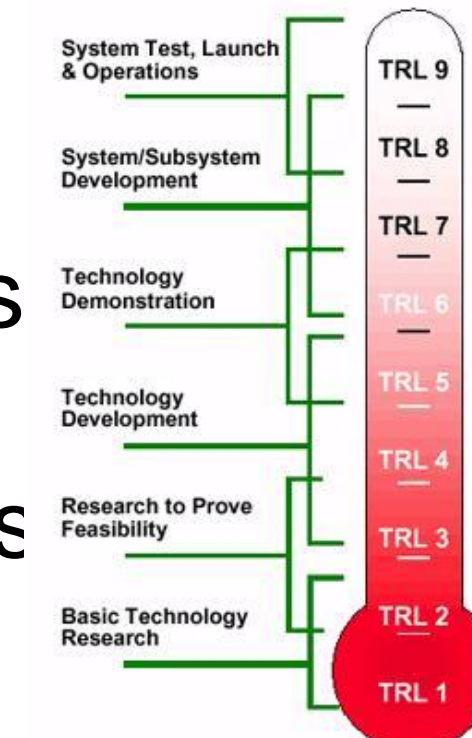


Cascade Cycle Hardware



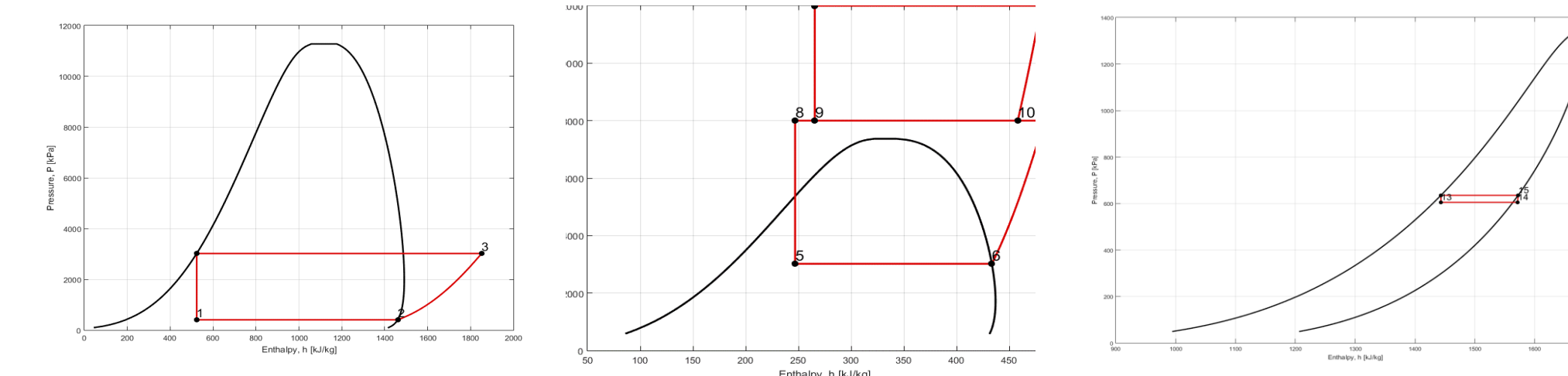
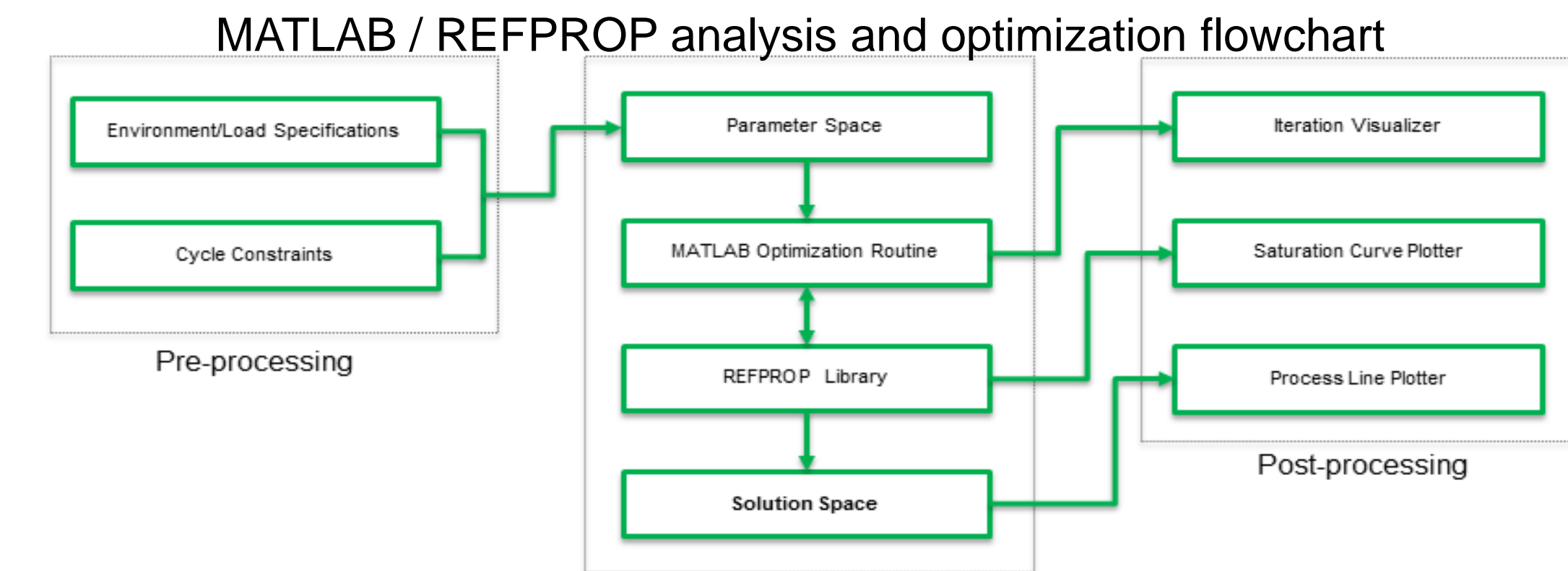
Technology Readiness

- NH3 components 6 < TRL < 9
- Transcritical CO2 components 6 < TRL < 7.5
- Supercritical CO2 components 6 < TRL < 7.5
- FAME / MLL components 1 < TRL < 3
- Need to research and develop FAME/MML compressor, seals, heat exchangers and throttling valve



CF. <http://as.nasa.gov/>

Cycle Optimization



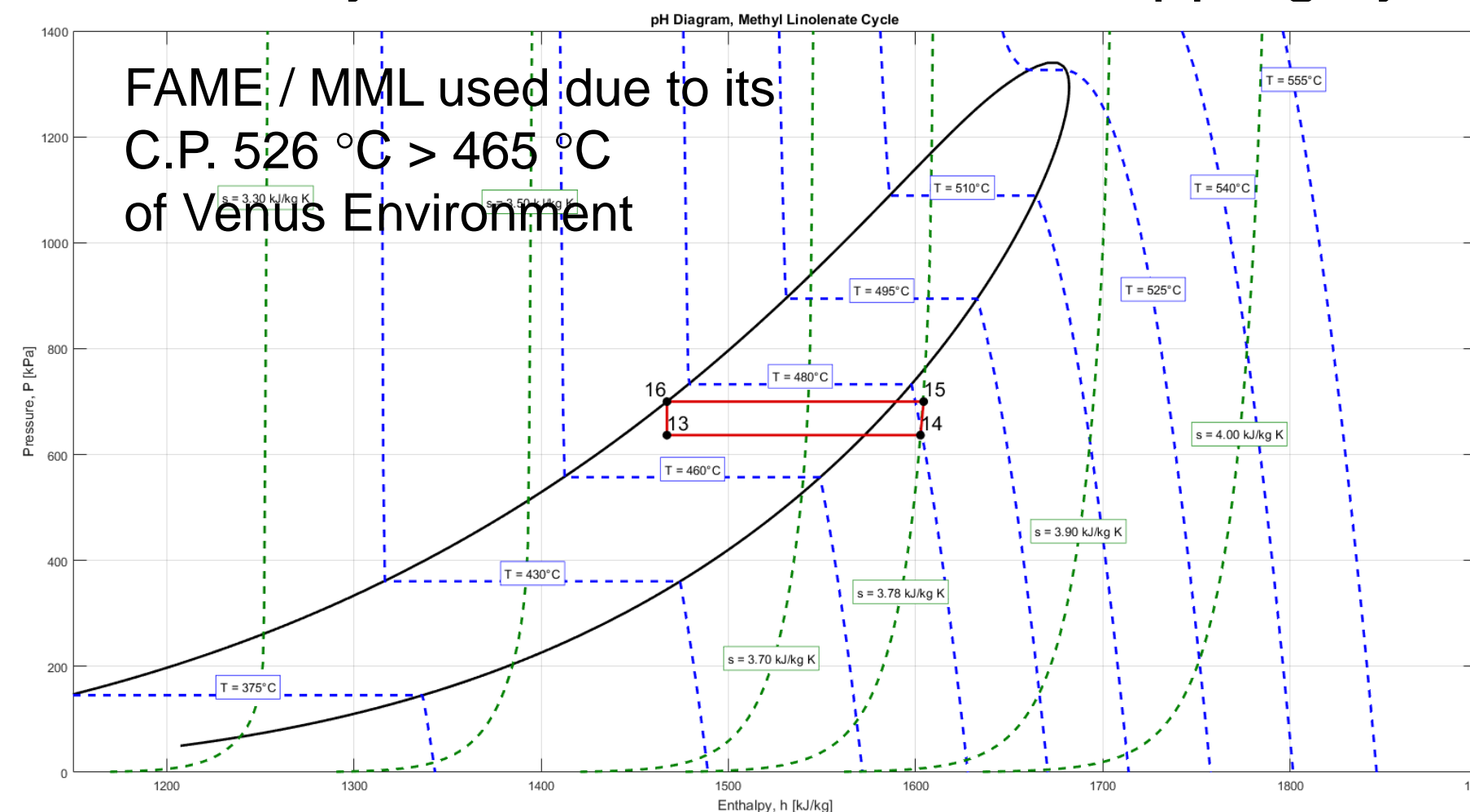
PH diagrams for NH3 / TSCO2/ SCO2 / FAME MLL optimized cycles

Results

Fluid	Flow rate (kg/hr)	Compressor Power (W)
NH3	384e-3	42
TCO2	2.73	38
SCO2	3.35	21
FAME / MLL	5.63	1.4

Overall system COP = 0.98 to lift 100 W of electronics power dissipation @ 100 °C

- Cascade cycle is novel in its use of traditional and supercritical fluids
- NH3 bottoming, CO2 middling and Fatty Acid Methyl Ester / Methyl Linoleate FAME/MML topping cycle



FAME / MML used due to its C.P. 526 °C > 465 °C of Venus Environment