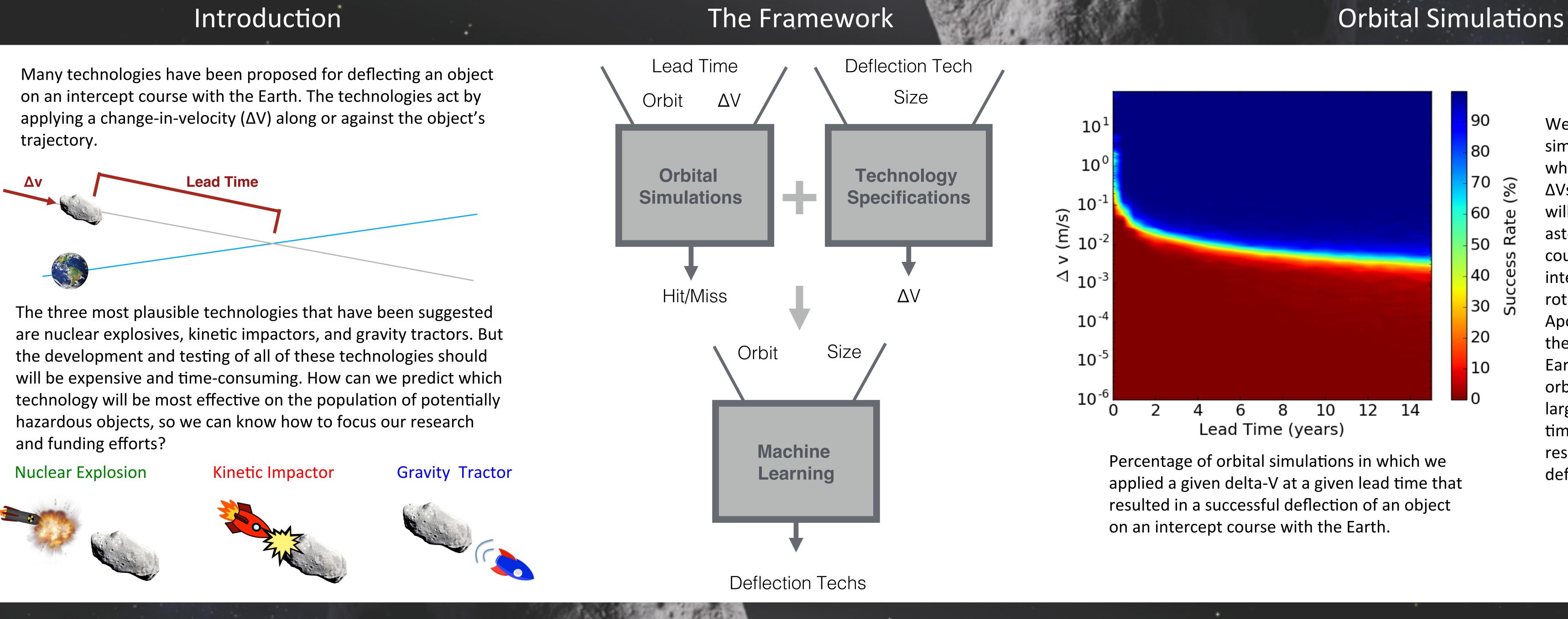
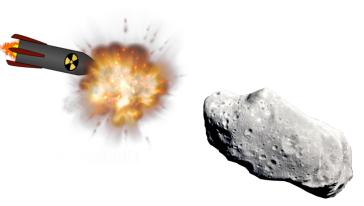


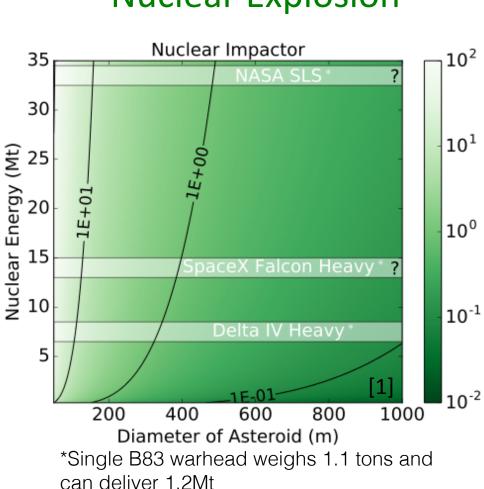
Erika Nesvold<sup>1,2</sup>, Nicolas Erasmus<sup>1,3</sup>, Adam Greenberg<sup>1,4</sup>, Elmarie van Heerden<sup>1,5</sup>, J.L. Galache<sup>6</sup>, Eric Dahlstrom<sup>7</sup>, Franck Marchis<sup>8</sup> For more information contact: Erika Nesvold (enesvold@carnegiescience.edu)





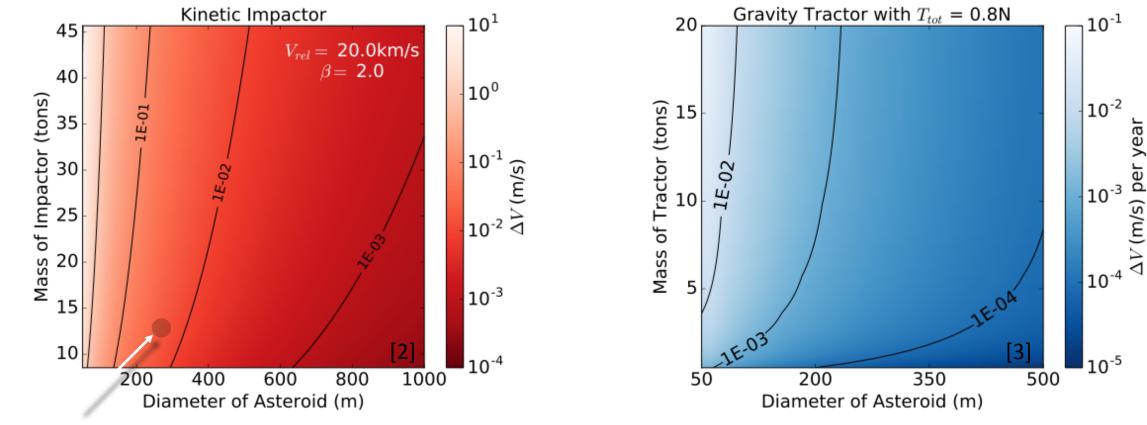


## **Technology Specifications**



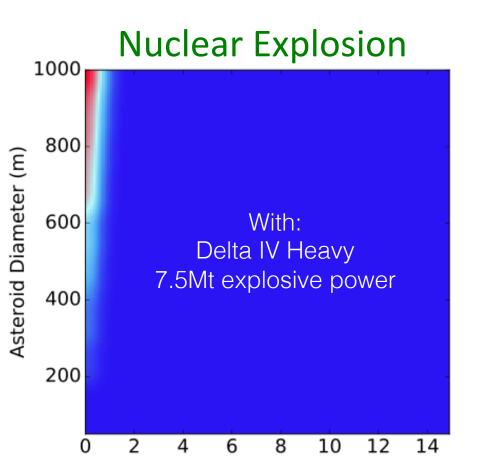
## Nuclear Explosion

## **Kinetic Impactor**

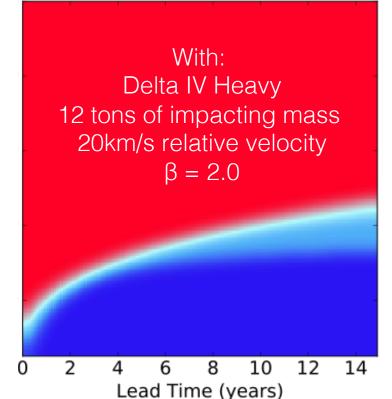


Delta IV Heavy hits 300m asteroid:  $\Delta V = ~13 \text{ mm/s}$ 

We then analyzed the predicted capabilities of the three most plausible deflection technologies to determine whether they could produce the necessary ΔV values (above). For a given object size and lead time, we can predict the likelihood of success for a given technology (below).



## **Kinetic Impactor**



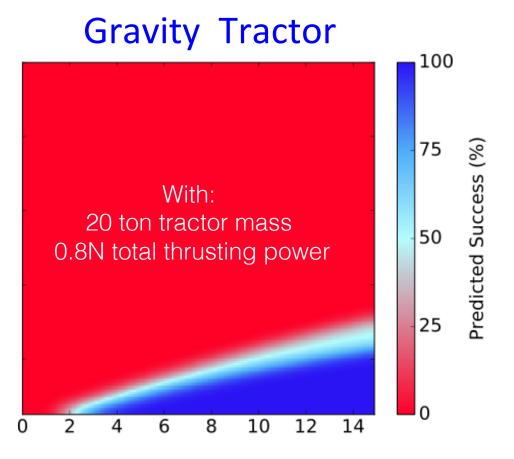
Calculations based on equations from:

[1] P. Hammerling et. al., Acta Astronautica, 36, 337-346, (1995) [2] J.P. Sanchez Cuartielles, PhD Thesis, (2009) [3] E. T. Lu et. al., Nature, 438, 177, (2005)

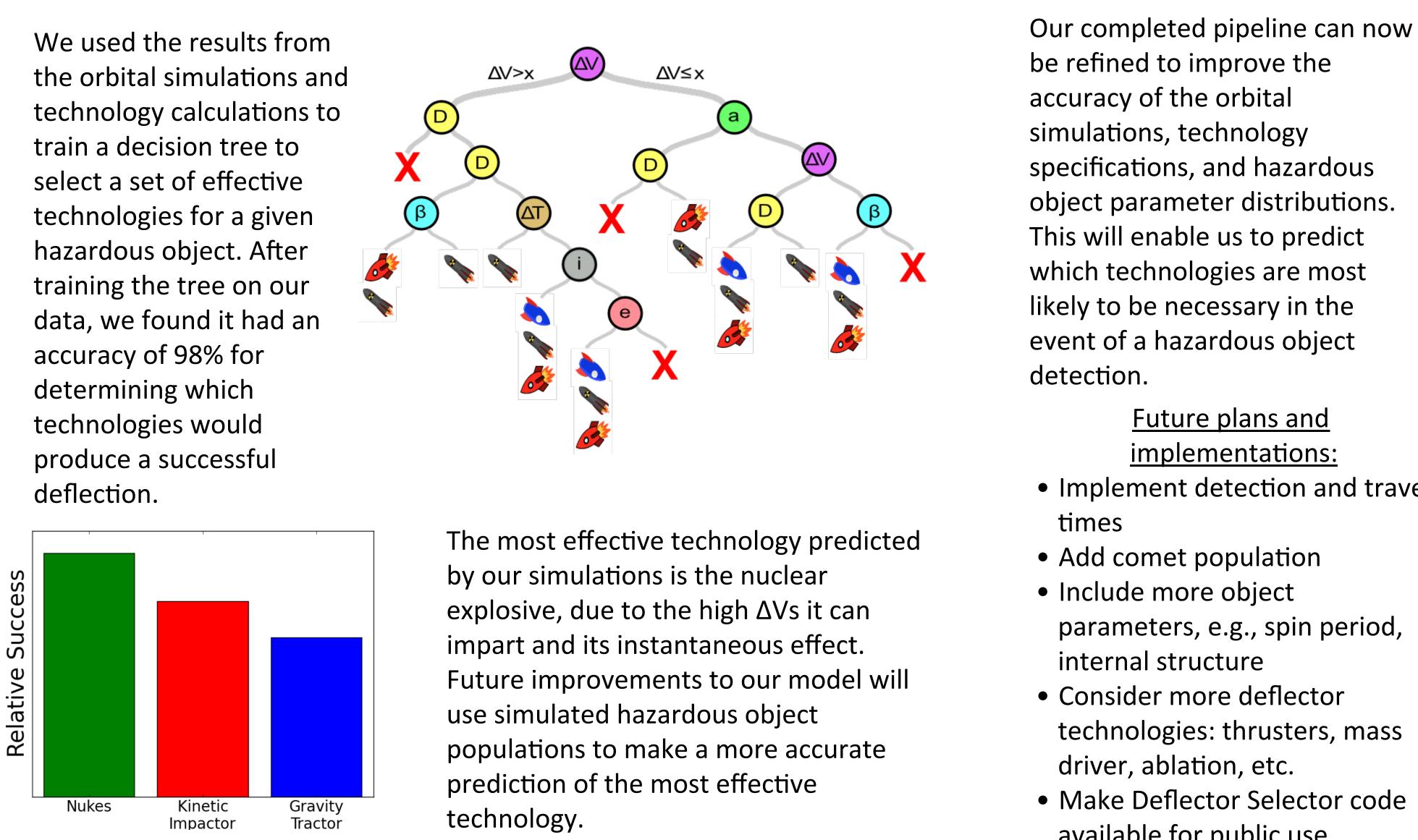
For more information on the NASA Frontier Development Lab, visit www.frontierdevelopmentlab.org or scan the QR code  $\rightarrow$ 

# The Deflector Selector: A Machine-Learning Framework for Prioritizing **Deflection Technology Development**

**Gravity Tractor** 







Institute



# Machine Learning

Affiliations: (1) NASA Frontier Development Lab, (2) Carnegie Institution Department of Terrestrial Magnetism, (3) South African Astronomical Observatory, (4) University of California, Los Angeles, (5) Oxford University, (6) IAU Minor Planet Center, (7) International Space Consultants, (8) SETI



We first used orbital simulations to test whether applying random  $\Delta Vs$  at random lead time will deflect a near-Earth asteroid on an intercept course. We generated the intercept orbits by rotating known Aten and Apollo asteroid orbits until they intercepted the Earth, without changing orbit shapes. As expected, larger  $\Delta Vs$  and longer lead times are more likely to result in a successful deflection.

## Conclusion

be refined to improve the accuracy of the orbital simulations, technology specifications, and hazardous object parameter distributions. This will enable us to predict which technologies are most likely to be necessary in the event of a hazardous object Future plans and implementations: Implement detection and travel Add comet population Include more object parameters, e.g., spin period, internal structure • Consider more deflector technologies: thrusters, mass driver, ablation, etc. Make Deflector Selector code available for public use

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